



# Overview of Micro/nano/pico-satellite

## Part 1

Shinichi Nakasuka  
University of Tokyo



CubeSat 03,05



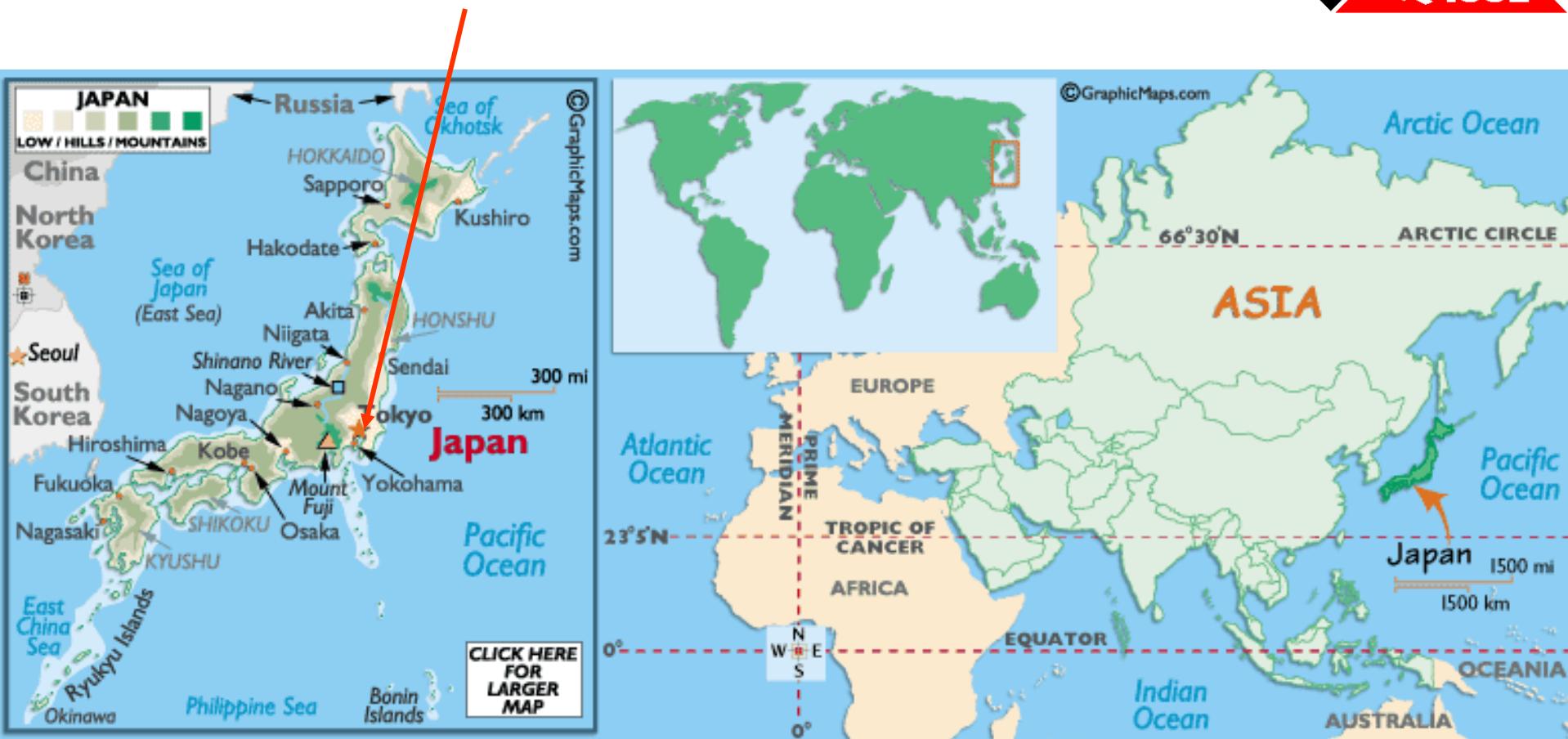
PRISM '09



Nano-JASMINE (TBD)

# Where is Our University ?

- University of Tokyo: 28,000 students, 7800 staffs
  - School of Engineering: 22 departments
    - Intelligent Space Systems Laboratory



# Statistics about University of Tokyo

- The University of Tokyo
  - Established in 1886
  - 11 Faculties/Graduate Schools
  - 14000 undergraduate/14000 graduate students
  - 3900 academic staffs/ 3900 supporting staffs
  - **World QS ranking: 23th**
- School of Engineering
  - 1050 undergraduate students/year
  - 2053 master student/1066 Ph.D candidate
    - Including 921 foreign students from 74 countries
  - 453 academic staffs/ 206 supporting staffs
  - Yearly budget: about \$ 250M
  - **World QS ranking: 8th**
    - **Mechanical/Aerospace: 9<sup>th</sup> ranking**



# Graduate School of Engineering

18 Departments, 2 Institutes, 7 Centers

- ◆ Civil Engineering
- ◆ Architecture
- ◆ Urban Engineering \*3
- ◆ Mechanical Engineering
- ◆ Aeronautics and Astronautics
- ◆ Precision Engineering
- ◆ Electrical Engineering and Information Systems
- ◆ Applied Physics
- ◆ Systems Innovation
- ◆ Materials Engineering
- ◆ Applied Chemistry
- ◆ Chemical System Engineering
- ◆ Chemistry and Biotechnology
- ◆ Advanced Interdisciplinary Studies \*1
- ◆ Nuclear Engineering and Management
- ◆ Bioengineering
- ◆ Technology Management for Innovation
- ◆ Nuclear Professional School \*2

- Institute of Engineering Innovation
- Institute of Innovation in Engineering Education (IIIIEE)
- Center for Water Environment Technology
- Quantum-Phase Electronics Center
- Frontier Research Center for Energy and Resources
- Photon Science Center
- Medical Device Develop. and Regulation Research Center
- Resilience Engineering Research Center
- Center for Spintronics Research Network

- ✓ \*1 Doctoral course only
- ✓ \*2 One year professional graduate course
- ✓ \*3 “Master’s Program in Sustainable Urban Regeneration” also offered

# Department of Aeronautics and Astronautics

- Department of A&A
  - Aerodynamics: 2 labs
  - Aircraft design: 1
  - Structure/material: 2
  - Aircraft Control: 1
  - Information Technology 1
  - Space Systems: 1
  - Engine 2
  - Electric propulsion 1
  - Combustion 1
  - Tribology 1
- Research Center for advanced Science and Technology (Univ. Tokyo)
  - Remote Sensing 1
  - AI and Space Robots 1
- ISAS (outside)
  - Aerodynamics 2
  - Material/Structure 2
  - Control/Systems 2
  - Propulsion/Engines 2

# Shinichi Nakasuka, Ph.D, Prof

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- Graduated from University of Tokyo in Aeronautics in 1983
- Received Ph.D in Aeronautics in 1988
- Joined IBM Research from 1988 to 1990
- Has been working for School of Engineering, University of Tokyo since 1990 as;
  - Lecturer: 1990-1992
  - Associate Professor: 1993-2004
  - Professor: 2004-now
- Space Policy Committee of Government

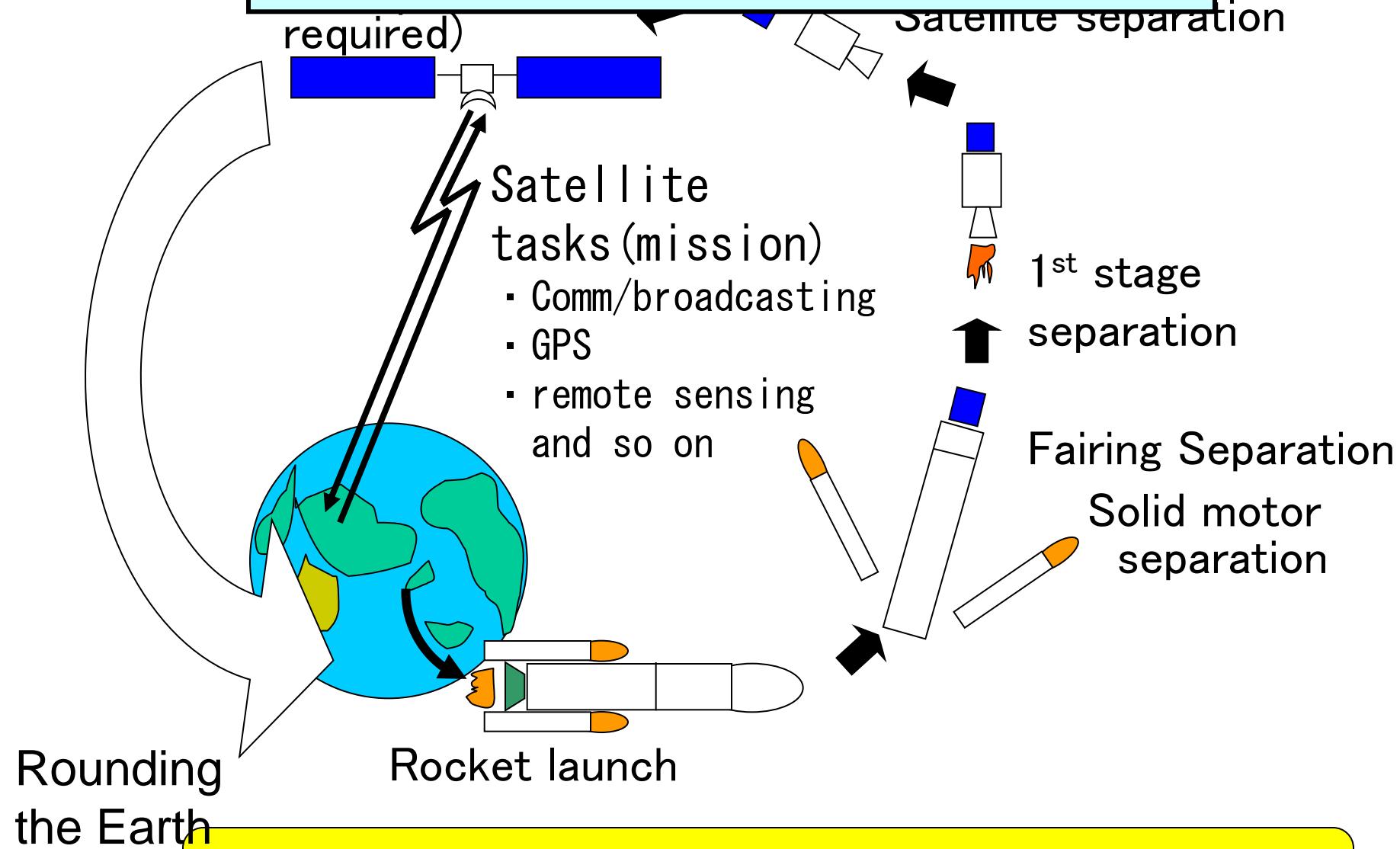
# Major Research Fields

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- Navigation, Guidance and Control of Spacecraft
- Micro/Nano/Pico Satellite Design, Development and Operation
- Novel Future Space Systems
- Autonomy and Intelligence for Space Systems
- Artificial Intelligence, Machine Learning

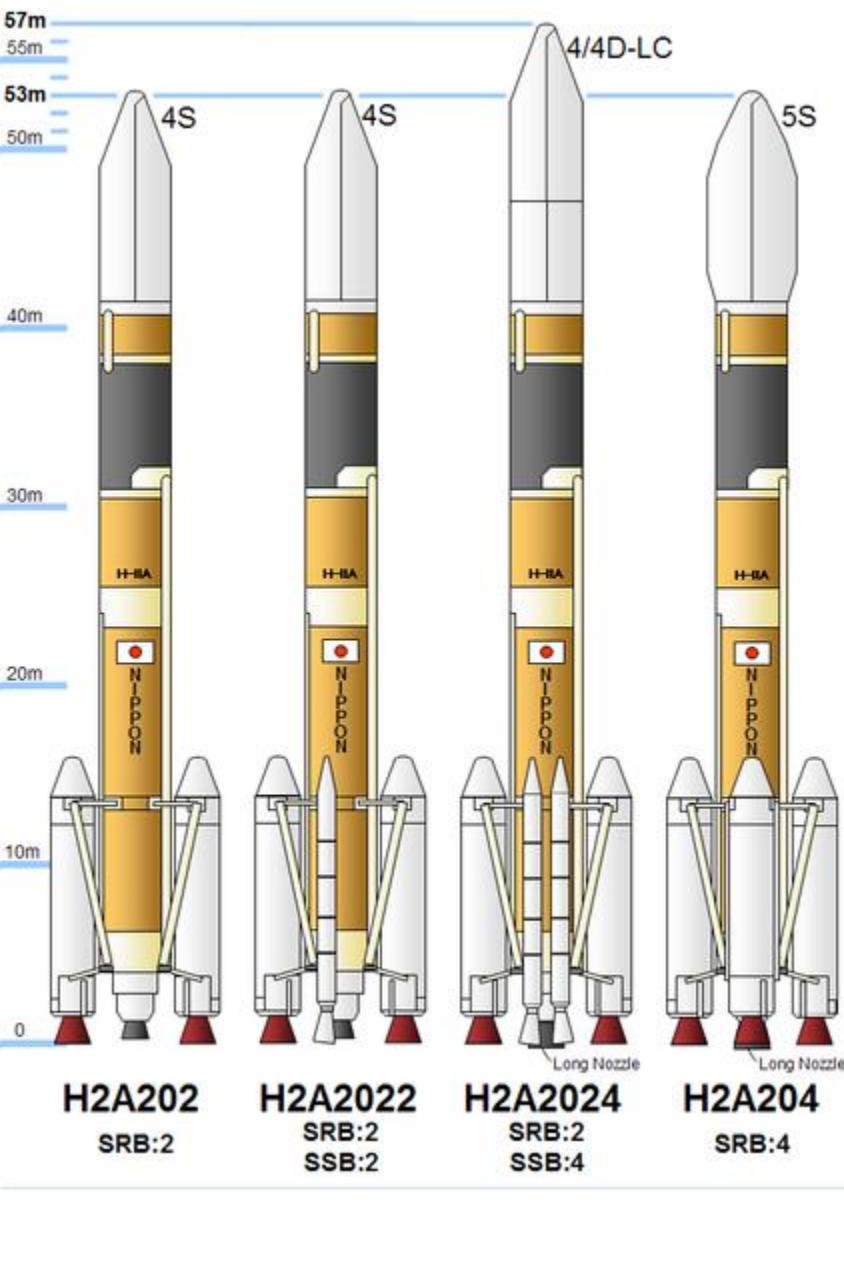
# Rocket and Satellite

# Rocket and Satellite



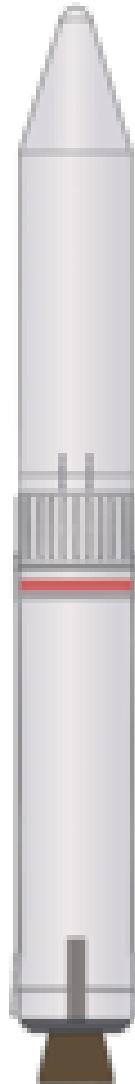
“Rocket” Carries “Satellites” to Orbits

# H-IIA 46/47



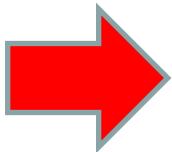
| Height                                | 53.5 m (173 ft)                             |                         |                                     |
|---------------------------------------|---|-------------------------|-------------------------------------|
| Diameter                              | 4 m (13.1 ft)                               |                         |                                     |
| Mass                                  | 285,000 - 445,000 kg (628,317 - 981,057 lb) |                         |                                     |
| Stages                                | 2   |                         |                                     |
| <b>Capacity</b>                       |   |                         |                                     |
| Payload to LEO                        | 10,000 - 15,000 kg (22,046 - 33,069 lb)     |                         |                                     |
| Payload to GTO                        | 4,100 - 6,000 kg (9,038 - 13,227 lb)        |                         |                                     |
| Designation                           | Mass (tonnes)                               | Payload (tonnes to GTO) | Add-on modules                      |
| H2A202                                | 285   | 4.1                     | 2 SRB-A (SRB)                       |
| H2A2022 (discontinued) <sup>[4]</sup> | 316   | 4.5                     | 2 SRB-A (SRB) + 2 Castor 4AXL (SSB) |
| H2A2024 (discontinued)                | 347   | 5                       | 2 SRB-A (SRB) + 4 Castor 4AXL (SSB) |
| H2A204                                | 445   | 6                       | 4 SRB-A (SRB)                       |
| H2A212 (cancelled)                    | 403   | 7.5                     | 2 SRB-A (SRB) + 1 LRB               |
| H2A222 (cancelled)                    | 520   | 9.5                     | 2 SRB-A (SRB) + 2 LRBS              |

# EPSIRON 4/4



| Development cost                | 20.5B YEN<br>E-X:5.3B YEN(1-3)             |
|---------------------------------|--|
| Cost/flight                     | E-X:3.8B YEN(4-)<br>E-I:< 3 B YEN          |
| Base                            | <u>H-IIA SRB-A</u> , <u>M-V</u> 3,4 stages |
| Features                        |  |
| Stages                          | 3 (base)<br>4 (option)                     |
| Booster                         | No   |
| Total weight                    | 90.8t                                      |
| Length                          | 24.4 m                                     |
| Diameter                        | 2.6 m                                      |
| Payload Capability              |  |
| <u>To LEO</u>                   | 1,200 kg<br>250 km × 500 km                |
| <u>To SSO</u> (Sun Synchronous) | 450 kg<br>500 km × 500 km                  |

# Future Rocket ?



Expendable  
→ Reusable

Cylinder type  
→ Winged type



# Manned Space Flight

# In 1961, when I was born, what happened in space ?

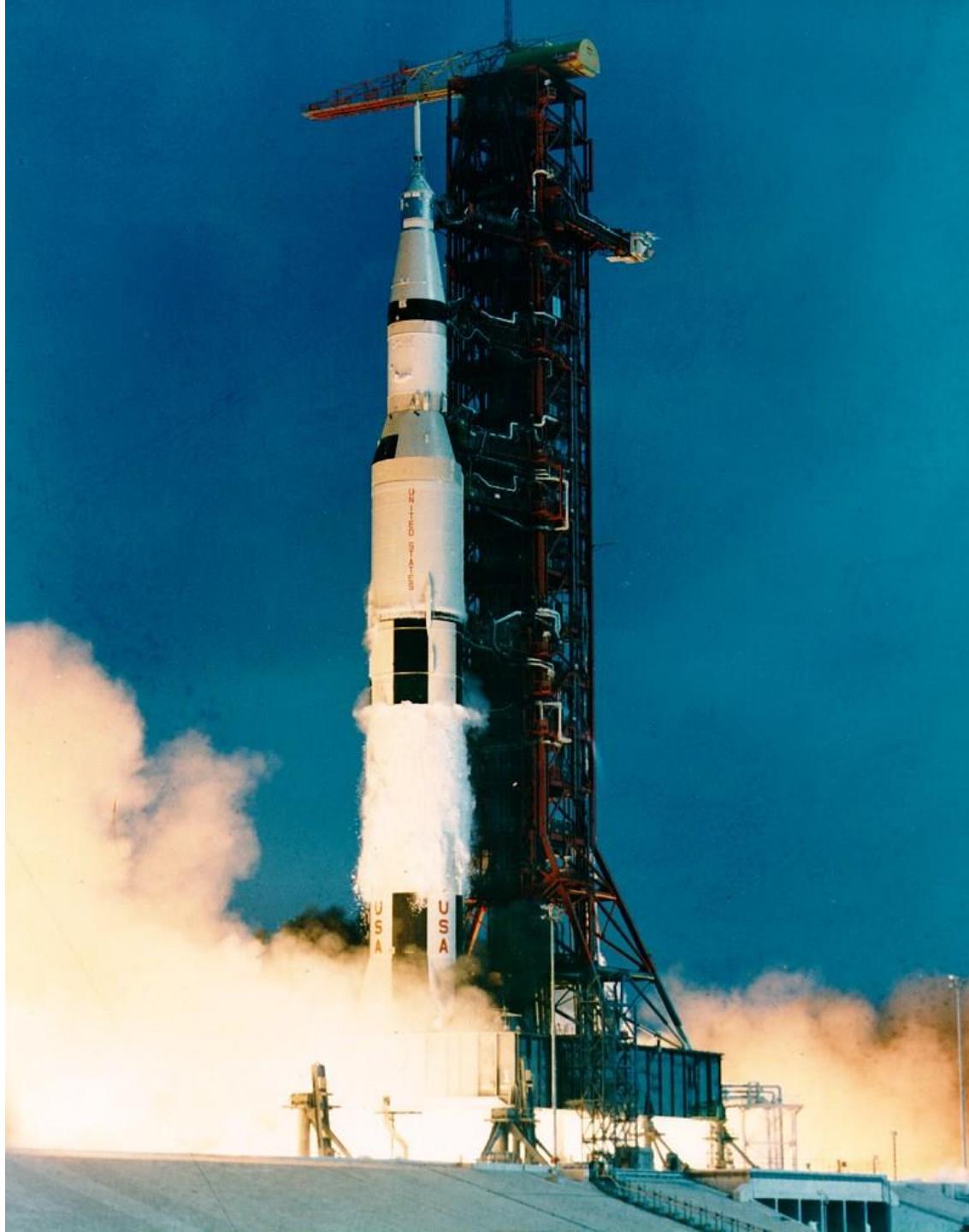
First human flight in space  
(April 12, Soviet's Yuri Gagarin)  
Vostok-1



Just 4 years after the successful launch of the world first satellite “Sputnik”

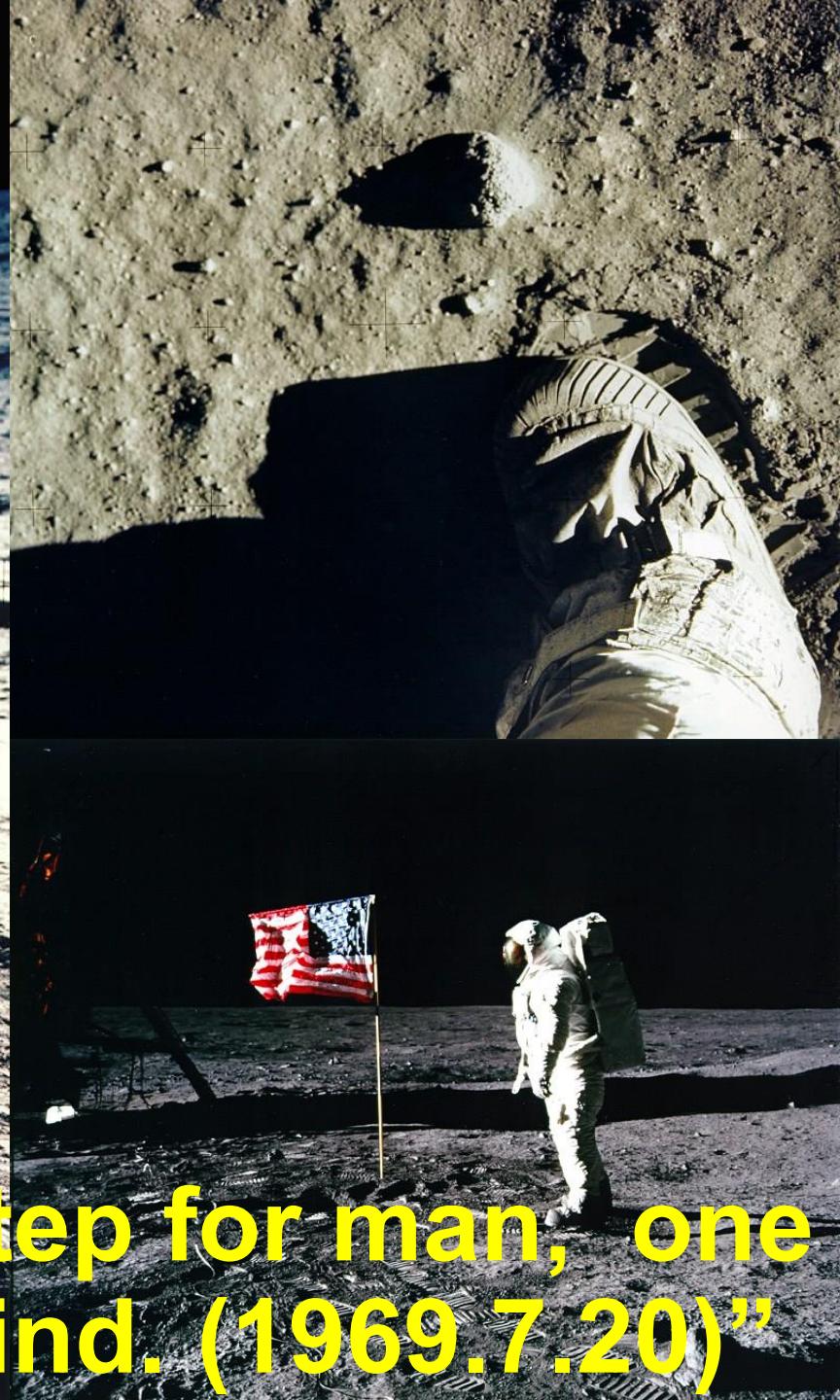
July 16,  
1969

Apollo 11



# Earth Rise from Moon Horizon





**“That's one small step for man, one  
giant leap for mankind. (1969.7.20)”**

# Japanese Lunar Exploration Satellite



3 ton

**KAGUYA (SELENE)**

Various sensors to  
observe the moon

# Manned Flight → Manned Stay in Space

Skylab (1973-74)   Mir (1986-2001)



# International Space Stations (ISS)

*- 5 principals and 15 partners -*



# Space Science and Exploration

# Space Telescope

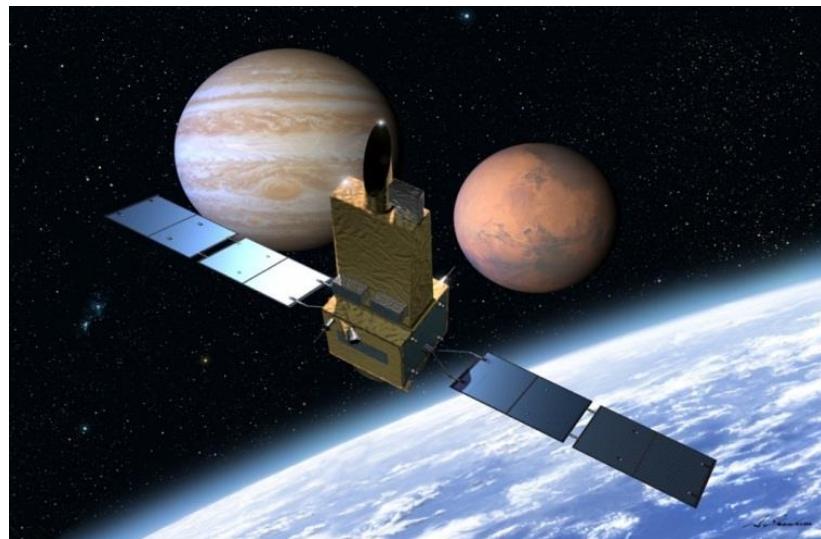


**Hubble Space Telescope (1990~)    Φ2.4m   11 t**

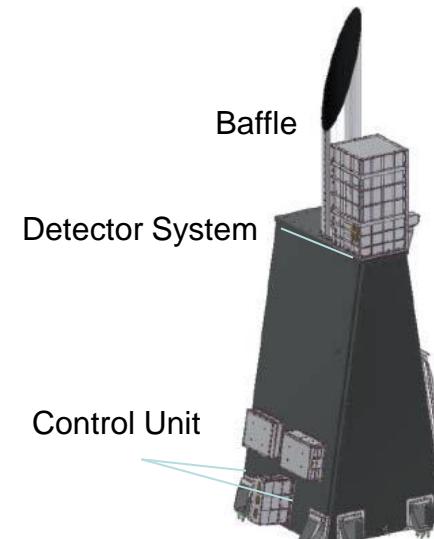
# M-Class (1): SPRINT-A/EXCEED (HISAKI)

Launched on 14 Sep. 2013 (335kg) on first Epsilon rocket launch

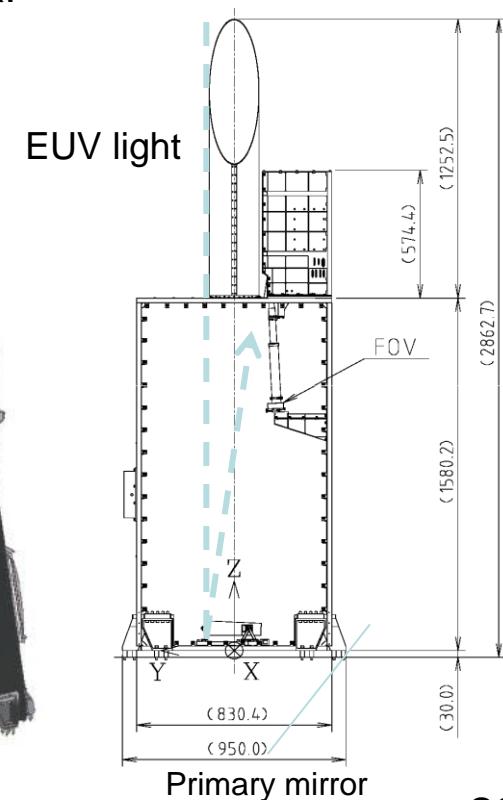
- Extreme ultraviolet spectrosCope for ExosphEric Dynamics
- The mission is spectroscopic and imaging observation of EUV (extreme ultraviolet: 60-145nm) emissions from tenuous plasmas around **Venus, Mars, Mercury, and Jupiter.**
  - measuring the plasma escape rates from the terrestrial planets (Venus, Mars, and Mercury)
  - understanding the electron energy and density distribution around the Jovian inner magnetosphere.



SPRINT-A/EXCEED



Mission Data Processor (MDP)



# M-class (2) ERG (Energization & Radiation in Geospace)

Launched in December 2016 (350kg), named “ARASE”

ERG is a mission to elucidate acceleration and loss mechanisms of relativistic electrons of Van Allen belts during space storms.



## Significance of the project:

- Direct observations on generation of relativistic electrons at the magnetic equator in the inner magnetosphere
  - contribution to understanding of the particle acceleration.
- Instrumental development to measure plasma and fields under the incidence of radiation belt particles with small satellite
  - contribution to a future Jovian mission.
- Understanding the acceleration and loss mechanisms.
  - contribution to predictable space weather model for space radiation environments.

# M-class (3) SLIM (Smart Lander onto the Moon)

Planned to be launched in FY2019 (520kg)

SLIM is a mission to demonstrate the technology for pin-point (about 100m accuracy) soft landing on lunar surface.

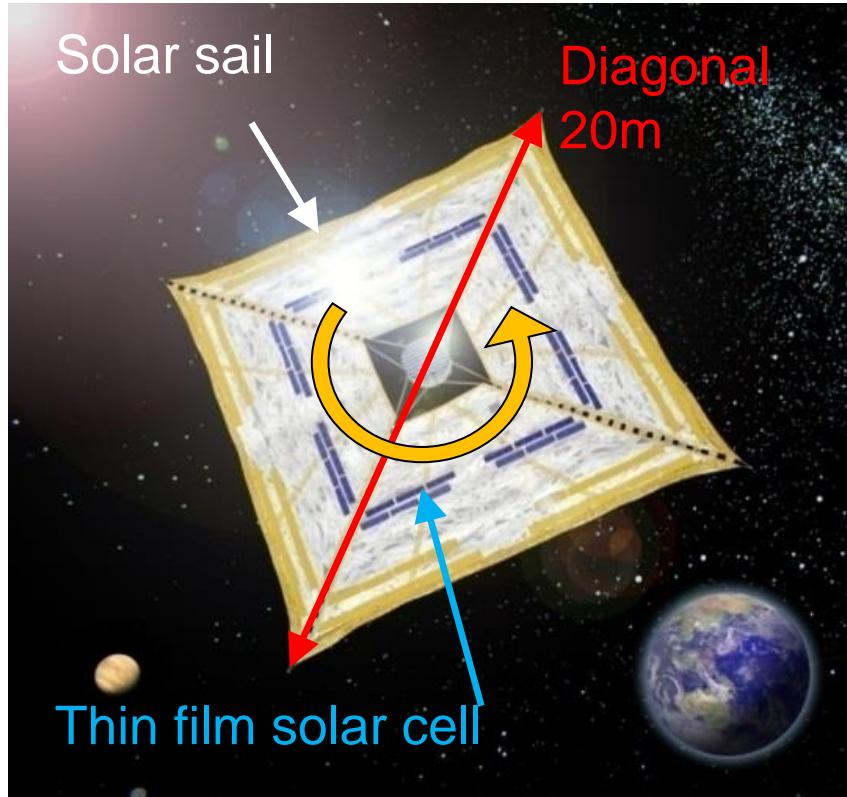


- Technology demonstration with Small Spacecraft
  - Image-based Navigation utilizing Lunar Terrain
  - Autonomous Obstacle Detection
  - Robust Pin-point Guidance
  - Landing Shock Absorber
  - High-performance Propulsion
  - Exploration using Spectrometer or Tiny Rovers (option)
- Frequent trials of lunar/planetary surface exploration technology
- Precursor of future full-scale lunar or planetary missions

# (Piggy-back) IKAROS: A Solar Sail Demonstrator

**IKAROS** (= Interplanetary Kite-craft Accelerated by Radiation Of the Sun) is the world's first interplanetary solar sail craft which demonstrated its photon sailing and thin film solar power generation

- (310kg, launch in 2010 as piggyback of Akatsuki)

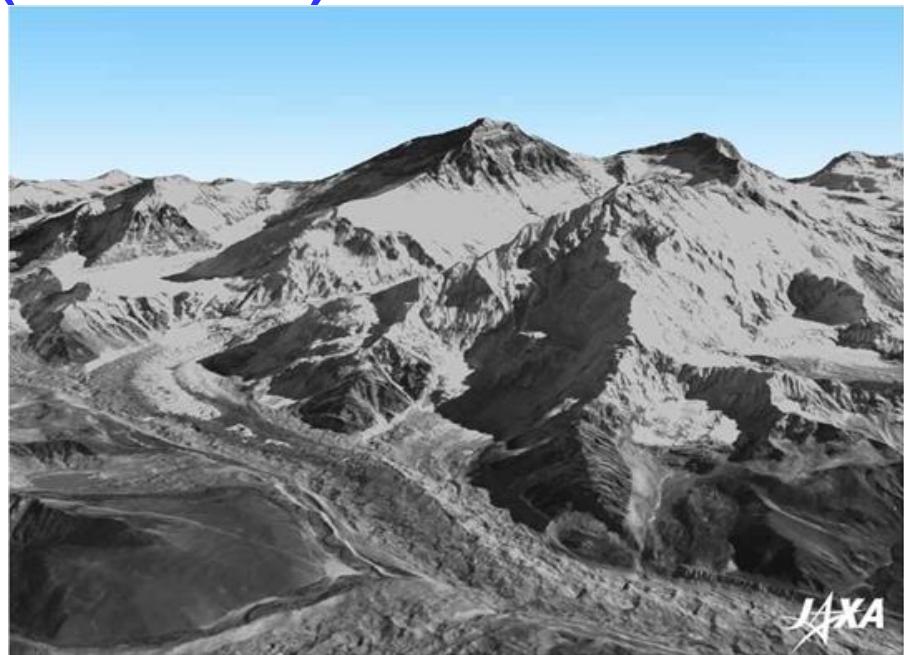
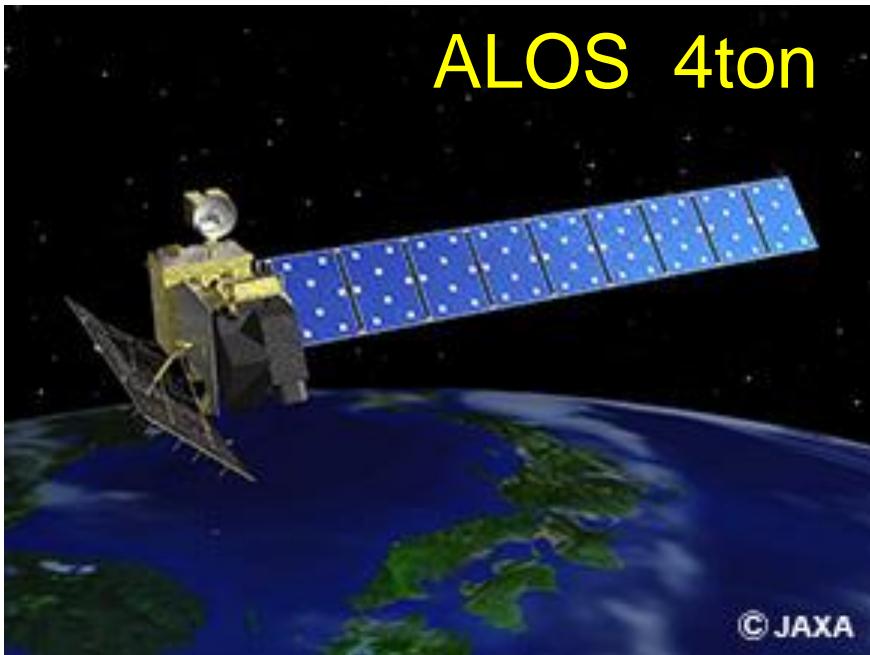


DCAM(Camera-craft) captured  
solar-sailing IKAROS  
June 14, 2010

# Governmental Satellites for Life on Earth

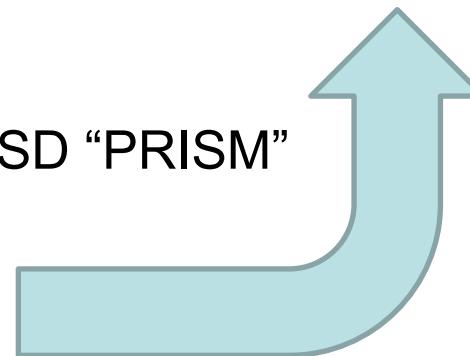
# Optical Remote Sensing Satellite - Daichi (ALOS) -

ALOS 4ton



DEM (Digital Elevation Map)

2.5m GSD “PRISM”  
Image



# ALOS-2 Synthetic Aperture Radar(SAR)

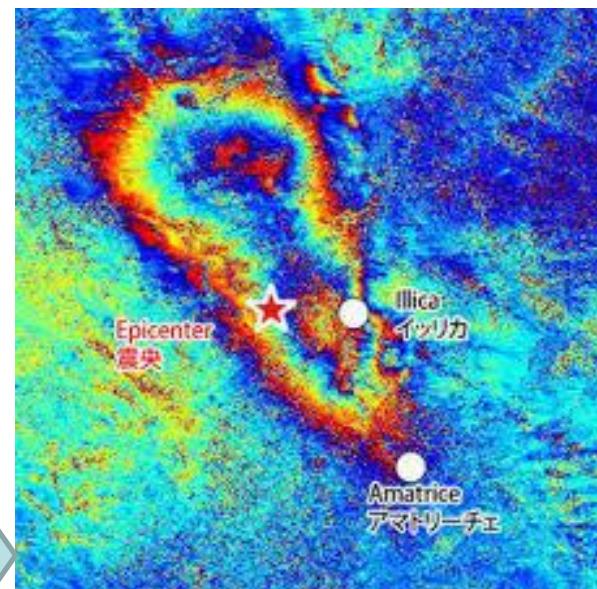


ALOS-2 2 ton



Ground info.  
can be  
captured  
under night  
or cloudy  
conditions

cm movement  
can be detected



# Meteorological/ATC Satellite

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**“HIMAWARI-8”  
(3.5 ton)**

# Navigation Satellite (GPS, QZS)



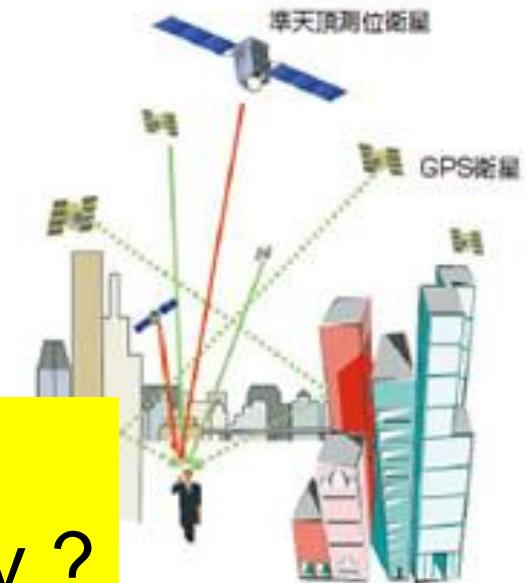
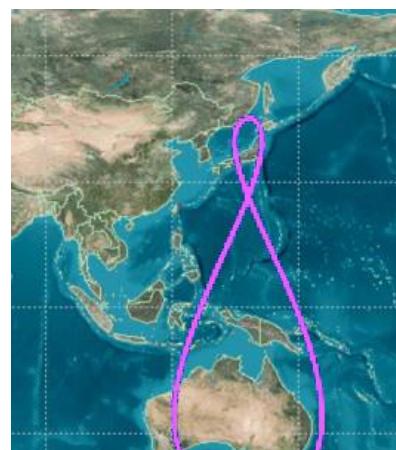
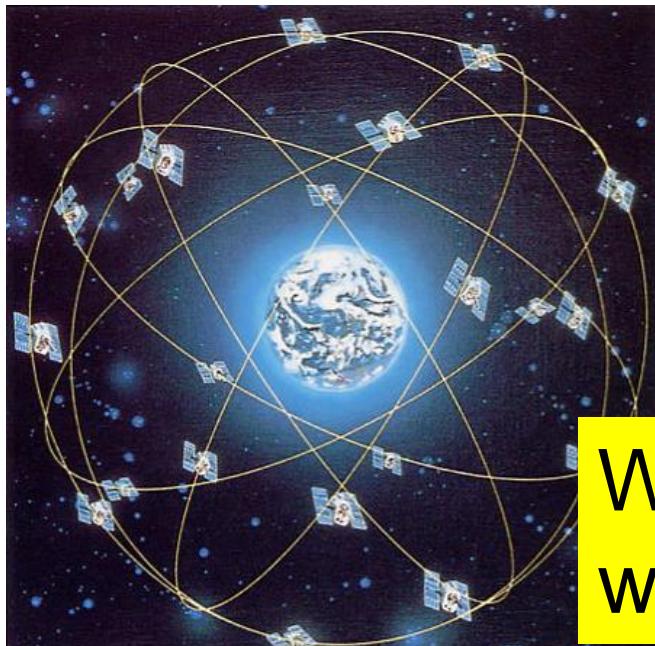
←GPS  
Accuracy  
3-5m



QZS→  
accuracy:  
**6cm**

- 1m  
(2018)

4ton



What you can do  
with 6 cm accuracy ?